1.1.

There are many possible ways to deduce this, but perhaps the most straightforward is to write Newton's law as $F = ma/g_c$ and see what numerical value and units are required to make it work out. If one poundal is the force required to accelerate 1 lb_m at 1 ft/s^2, then Newton's law in these units is: 1 poundal = 1 lb_m * 1 ft/s^2 / g_c. So, in these units, $g_c = 1$ (lb_m ft)/(poundal s^2).

1.1 (7th edition Prob. 1.2)

(*a*) Power is *power*, whether it is electrical, mechanical, or otherwise. Thus, electric power has the usual units of power:

$$\text{power} = \frac{\text{energy}}{\text{time}} = \frac{\text{J}}{\text{s}} = \frac{\text{N m}}{\text{s}} = \frac{\text{kg m}^2}{\text{s}^3}$$

(*b*) Electric current is by definition the time rate of transfer of electrical charge. Thus

$$\text{current} = \frac{\text{charge}}{\text{time}}$$

or   charge = current*time = A s

(you probably recall that the Coulomb is the usual derived unit of charge, defined as 1 A s)

(*c*) Since power is given by the product of current and electric potential,

$$\text{power} = \frac{\text{energy}}{\text{time}} = \text{current*electric potential}$$

or   electrical potential $= \dfrac{\text{energy}}{\text{current*time}} = \dfrac{\text{kg m}^2}{\text{A s}^3}$

(you probably recall that this is defined as the volt)

(*d*) Because (by Ohm's Law) current is electric potential divided by resistance,

$$\text{current} = \frac{\text{electrical potential}}{\text{resistance}}$$

or   resistance $= \dfrac{\text{electrical potential}}{\text{current}} = \dfrac{\text{kg m}^2}{\text{A}^2 \text{ s}^3}$

(this is defined as the ohm)

(*e*) Since electric potential is electric charge divided by electric capacitance,

$$\text{electrical potential} = \frac{\text{charge}}{\text{electrical capacitance}}$$

or $$\text{electrical capacitance} = \frac{\text{charge}}{\text{electrical potential}} = \frac{\text{A s}}{\dfrac{\text{kg m}^2}{\text{A s}^3}} = \frac{\text{A}^2 \text{ s}^4}{\text{kg m}^2}$$

1.2 (7th edition Prob. 1.3)

We must convert both the units and the logarithm (between base 10 and natural logarithm). We know that t in degrees Celsius is equal to T in Kelvins minus 273.15. Also 1 kPa is equal to 7.50 torr (we might have to look up this conversion factor). So, we have

$$P^{Sat}/\text{torr} = 10^{\left(a - \frac{b}{t/°C + c}\right)} = 7.5 \bullet P^{Sat}/\text{kPa} = 7.5 \exp\left(A - \frac{B}{T/\text{K} + C}\right)$$

Next, we might recognize that 10 can be written as exp(ln(10)) or exp(2.303). That is how we convert from base 10 log to natural log in general. So,

$$\exp\left(2.303\left(a - \frac{b}{T/\text{K} - 273.15 + c}\right)\right) = 7.5 \exp\left(A - \frac{B}{T/\text{K} + C}\right)$$

Here, I have also substituted $T - 273.15$ for t. Taking the natural log of both sides gives

$$2.303\left(a - \frac{b}{T/\text{K} - 273.15 + c}\right) = \ln(7.5) + A - \frac{B}{T/\text{K} + C}$$

For the two functions to be equal for all values of T, each part of the functions must be the same, so we must have

$A = 2.303a - \ln(7.5)$ or $A = \ln(10)*a - \ln(7.5)$

$B = 2.303b$ or $B = \ln(10)*b$
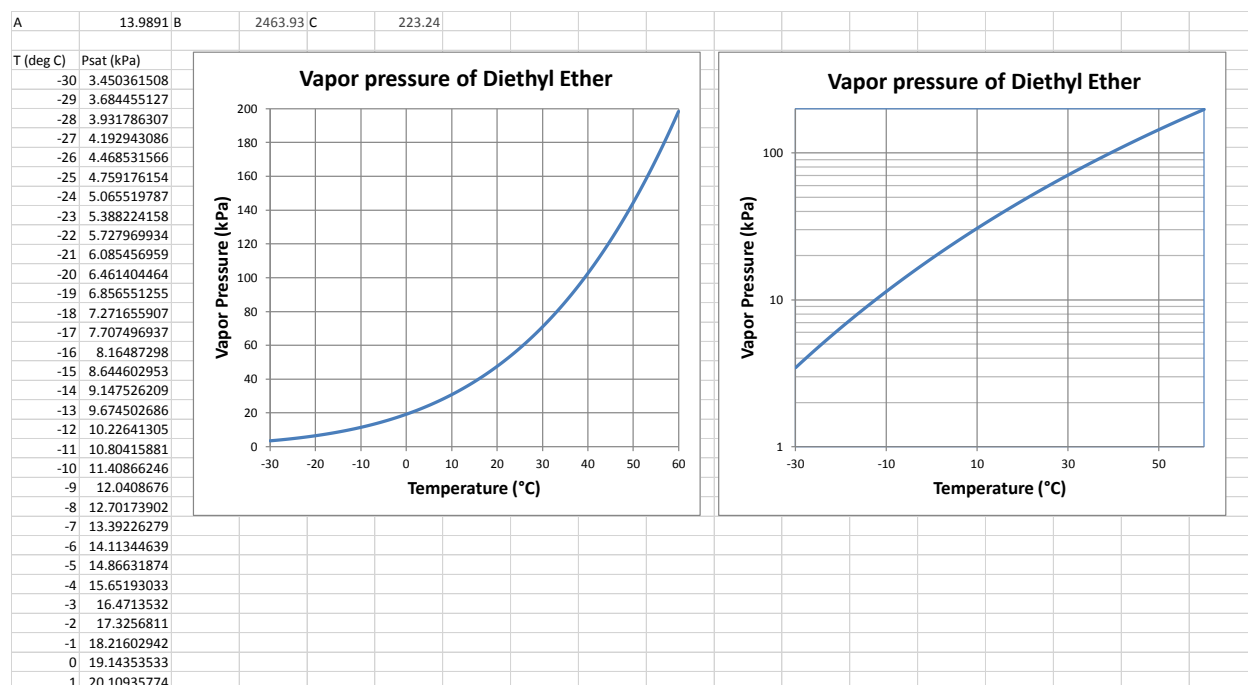
$C = c - 273.15$

1.3 (New)

The point of this problem is just for you to practice evaluating and plotting a simple function. You will do a number of problems over the course of the semester (and many more over the course of your career) in which the results are best presented in graphical form. The only thing to be careful of in plotting the Antoine equation is to pay attention to the units of T and P^{sat} and to whether the constants are given for use with the base 10 logarithm or the natural logarithm. The parameters in Table B.2 are for use with T in °C, P in kPa, and the natural logarithm.

I used MS Excel to prepare the plots for diethyl ether. A portion of the spreadsheet (containing the plots) is shown below:

A	13.9891	B	2463.93	C	223.24

T (deg C)	Psat (kPa)
-30	3.450361508
-29	3.684455127
-28	3.931786307
-27	4.192943086
-26	4.468531566
-25	4.759176154
-24	5.065519787
-23	5.388224158
-22	5.727969934
-21	6.085456959
-20	6.461404464
-19	6.856551255
-18	7.271655907
-17	7.707496937
-16	8.16487298
-15	8.644602953
-14	9.147526209
-13	9.674502686
-12	10.22641305
-11	10.80415881
-10	11.40866246
-9	12.0408676
-8	12.70173902
-7	13.39226279
-6	14.11344639
-5	14.86631874
-4	15.65193033
-3	16.4713532
-2	17.3256811
-1	18.21602942
0	19.14353533
1	20.10935774

Vapor pressure of Diethyl Ether

Vapor pressure of Diethyl Ether

1.4 (7th edition Prob. 1.4)

One way to solve this problem is to write both the Celsius and Fahrenheit temperatures in terms of the absolute temperature in Kelvin, then set these equal to each other and solve for the temperature in Kelvin. The Celsius temperature is T (°C) = T (K) – 273.15. The Fahrenheit temperature is T (°F) = 1.8*T (K) – 459.67. If we plot these two lines, they look like:

SVNAS, problem 1.4
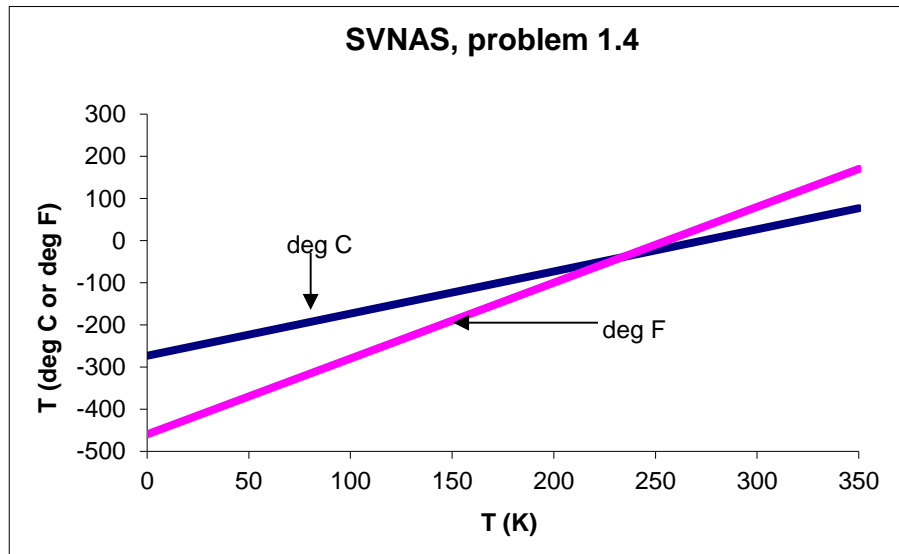
What we are asked to find is the point where these two lines cross. That is, where

$$T\,(K) - 273.15 = 1.8*T\,(K) - 459.67$$

Solving this for T (K) gives T (K) = (459.67 – 273.15) / 0.8 = 233.15 K. At this temperature (in K) the Fahrenheit temperature is 1.8 * 233.15 – 459.67 = -40 °F. Likewise, the Celsius temperature is 233.15 – 273.15 = -40 °C.

1.5 (New)

I found the following on www.bulbs.com. I chose all lamps that could be screwed into a regular medium-base socket

Type	Watts Used	Lumens Produced	Luminous Efficacy
Incandescent	150 W	1735 lm	12 lm W⁻¹
Halogen	120 W	1850 lm	15 lm W⁻¹
High Pressure Sodium	35 W	2300 lm	66 lm W⁻¹
Compact Fluorescent	23 W	1650 lm	72 lm W⁻¹
LED	19 W	1680 lm	88 lm W⁻¹

In this light-output range, the high pressure sodium lamp, compact fluorescent, and LED are comparable. All three high-efficiency options are about a factor of 4 better than incandescent or halogen bulbs. Eventually, LED's have the potential to be much more efficient than fluorescents, but what is on the market now is not there yet. They are improving rapidly – this is the first year in which the LED lamp has been the most efficient when I did this exercise. Previously, the high-pressure sodium lamp was most efficient.

1.6 (7th edition Prob. 1.5.)

Pressure is force divided by area. So, the maximum force required (to be applied by the weights) is the maximum pressure to be measured times the area of the piston. The area of the piston is

$$A = \pi d^2/4 = 4\pi \text{ mm}^2 = 12.57 \text{ mm}^2 = 1.257 \times 10^{-5} \text{ m}^2$$

The maximum pressure to be measured is 3000 bar = 3×10^8 Pa = 3×10^8 N m^{-2}. Thus, the maximum force to be applied by the weights is $F = 1.257 \times 10^{-5}$ m^2 * 3×10^8 N m^{-2} = 3770 N. If this force is to be applied by weights, and the acceleration of gravity is 9.8 m s^{-2}, then we have

$$F = mg = 9.8 \text{ m s}^{-2} * m = 3770 \text{ N} = 3770 \text{ kg m s}^{-2}$$

So, the mass of weights needed is approximately 3770/9.8 = 385 kg

1.7. (7th edition Prob. 1.6)

Pressure is force divided by area. So, the maximum force required (to be applied by the weights) is the maximum pressure to be measured times the area of the piston. The piston area is

$$A = \pi d^2/4 = \pi * 0.17^2/4 \text{ in}^2 = 0.0227 \text{ in}^2$$

The maximum pressure to be applied is 3000 atm = 3000 * 14.7 psi/atm = 44090 psi = 44090 lb$_f$ in^{-2}. For the standard acceleration of gravity (g numerically equal to g_c) one lb$_m$ imparts one lb$_f$ to

the piston. Thus, the mass of weights required is 0.0227 in² * 44090 lb_f in⁻² * (g/g_c = 1 lb_f/lb_m)⁻¹ = 1001 lb_m.

1.8 (7th edition Prob. 1.7)

The total pressure is equal to atmospheric pressure plus the weight per area (ρgh) of the mercury column. This is, in SI units, p = 101780 Pa + 13534 kg/m³ * 9.832 m s⁻² * 0.5638 m = 176800 Pa = 176.8 kPa.

1.9 (7th edition Prob. 1.8)

The total pressure is equal to atmospheric pressure plus the weight per area ($\rho(g/g_c)h$) of the mercury column. The only trick is to do this in the non-SI units given. Let us compute the pressure in psia (pounds-force per square inch, absolute). To do so, we should convert the density of mercury from g cm⁻³ to lb_m in⁻³. Thus, ρ = 13.543 g cm⁻³ * (2.54 cm/in)³ / 453.59 g/lb = 0.4893 lb_m in⁻³. We also need to convert atmospheric pressure from in Hg to psia by multiplying by 0.4912 psi/(in Hg). So, p = 29.86 in Hg * 0.4912 lb_f in⁻² (in Hg)⁻¹ + 0.4893 lb_m in⁻³ * 32.243 ft s⁻² / (32.1740 lb_m ft lb_f^{-1} s⁻²) * 25.62 in = 27.2 lb_f in⁻² = 27.2 psia.

1.10 (New)

The pressure unit "inches of water" is defined as the pressure exerted by one inch of water under standard gravitational conditions (density of water = 1 g/cm³, g = 9.8 m/s²). The pressure exerted on the gauge is given by

$$P = P_{atm} + \rho gh$$

Here P is the absolute pressure at the position of the gauge, ρ is the density of the water, g is the acceleration of gravity (~9.8 m/s²), and h is the depth of the water (50 m). We can take P_{atm} =

1.013 bar = 1.013×10^5 Pa. The pressure difference caused by the mass of the column of water above the pressure gauge (excluding atmospheric pressure) is then $P - P_{atm} = 6.064 - 1.013 = 5.051$ bar = 5.051×10^5 Pa. Converting that value to inches of water using my calculator gives 2028 inches of water. This is much closer to the actual depth, but still about 2.5% higher. This remaining discrepancy can be attributed to the density of seawater. If we take $h = 50$ m and $g = 9.8$ m/s^2, then we can use our pressure measurement to estimate the density of the seawater: $\rho = (P - P_{atm})/(gh) = 5.051 \times 10^5$ N/m^2/(9.8 m/s^2*50 m) = 1030.8 kg*m/(m s^2)/(m^2/s^2) = 1030.8 kg/m^3. This is a typical value for the density of seawater, which is higher than the density of fresh water because of its dissolved salt content.

1.11 (7th edition Prob. 1.9)

The advantages of spherical tanks stem from the fact that a spherical container has the minimum surface area for a given interior volume. Because of this:

(*a*) A minimum quantity of metal is required for tank construction (for a given metal thickness).

(*b*) The tensile stress within the tank wall is everywhere uniform, with no sites of stress concentration, and the stress is purely tensile (everywhere tangential to the tank surface). Moreover, the maximum stress within the tank wall is kept to a minimum.

(*c*) The surface area for heat transfer is minimized, optimizing the efficiency of cooling or insulating the tank.

1.12 (7th edition Prob. 1.10)

We know that 1 bar is equal to 750 torr or 750 mm of Hg or 0.75 meters of mercury. So to measure a pressure of 400 bar, he needed a column of mercury 400 bar * 0.75 m Hg/bar = 300 m of mercury.

1.13 (7th edition Prob. 1.11)

The spring constant can be determined from a force balance on the mass on earth. The gravitational force is $F = mg = 0.40$ kg * 9.81 m s^{-1} = 3.924 kg m s^{-1} = 3.924 N.

The spring force is $F = kx$, where k is the spring constant and x is the distance it is extended from its equilibrium length. These forces must be equal, so we have

$$kx = 3.924 \text{ N} = k * 1.08 \text{ cm} = k * 0.0108 \text{ m}.$$

So, $k = 3.924/0.0108 = 363.3 \text{ N/m}.$

Now, we do the same force balance on Mars, with g (for Mars) as the unknown.

$$F = kx = 363.3 \text{ N/m} * 0.40 \text{ cm} = 363.3 \text{ N/m} * 0.004 \text{ m} = 1.453 \text{ N}.$$

and $F = 1.453 \text{ N} = mg = 0.40 \text{ kg} * g$

So, $g = 3.63 \text{ N/kg} = 3.63 \text{ m s}^{-2}.$

1.14 (7th edition Prob. 1.12)

Substituting the ideal gas expression for density into the equation for variation of fluid pressure with height gives

$$\frac{dP}{dz} = -\frac{MP}{RT}g$$

If T is constant (along with M and g) then this is a separable equation that can be rearranged to give:

$$\frac{dP}{P} = -\frac{Mg}{RT}dz$$

This can be directly integrated to give

$$\ln\left(\frac{P}{P_o}\right) = -\frac{Mg}{RT}(z - z_o)$$

Or

$$P = P_o \exp\left(-\frac{Mg}{RT}(z - z_o)\right)$$

The standard pressure at sea level is $P_o = 101325$ Pa. If Denver is 1 mile $= 1609$ m above sea level, $g = 9.8 \text{ m s}^{-2}$, $M = 0.029$ kg/mol, $T = 10 \,°C = 283$ K, and $R = 8.314 \text{ kg m}^2 \text{ mol}^{-1} \text{ K}^{-1}$, then

$$P = 101325 \exp\left(-\frac{0.029 * 9.8}{8.314 * 283}(1609)\right) = 83430 \text{ Pa}$$

The estimated pressure is 83430 Pa = 0.834 bar = 0.823 atm.

1.15 (7th edition Prob. 1.13)

The force on the scale, whether on earth or on the moon, can be written at $F = mg/g_c$, and it is the force that is actually measured by the scale. The reading (displayed in lb_m) is therefore proportional to the actual mass and proportional to the gravitational constant., It can be written as $m_{read} = A\, m_{true}$, where A, the calibration constant for the scale, is numerically equal to g/g_c. On earth, where $g = 32.186\ \text{ft/s}^2$, $A = 1$. On the moon, where $g = 5.32\ \text{ft/s}^2$, A needs to be corrected by the ratio of g on the moon to g on earth. $A = 5.32/32.186 = 0.1653$. So, on the moon, $m_{read} = 0.1653 * m_{true}$. The true mass of the rock is $m_{true} = m_{read}/0.1653 = 18.76/0.1653 = 113.5\ \text{lb}_m$. Its *weight* on the moon is numerically equal to the reading. It is $18.76\ \text{lb}_f$.

1.16 (New)

(a) "Blood pressure" refers to the pressure exerted by the blood on the interior walls of the major arteries. The higher number (systolic pressure) represents the maximum pressure during the cardiac cycle, and the lower number (diastolic pressure) represents the minimum pressure during the cardiac cycle. In certain situations where patients require continuous, accurate, and rapid blood pressure monitoring, it is measured directly. This can be done by simply inserting a cannula (big needle) into a large artery and connecting it to a pressure transducer. This is much the same way that one measures pressure in a piece of chemical process equipment. Ordinarily, the blood pressure is measured indirectly using a sphygmomanometer. This is the familiar device with the inflatable cuff. What is measured is the pressure inside the cuff that is needed to stop the flow of blood through the artery it is squeezing (usually the brachial artery in the upper arm). The systolic pressure is the lowest pressure at which flow is completely stopped throughout the cardiac cycle, and the diastolic pressure is the pressure at which the flow is not stopped at any point in the cardiac cycle. Blood flow is detected by a health care practitioner listening for it using a stethoscope, or by a small transducer built into the inflatable cuff.

(b) Although they are often omitted, the units for blood pressure are mm of mercury. In SI units, a blood pressure of 120 over 80 would be a systolic pressure of 16000 Pa over 10700 Pa.

(c) It is a gauge pressure, measured and reported as the difference in pressure between the pressure in the blood vessels and the ambient pressure. For example, when a scuba diver descends 30 feet below the surface, her blood pressure remains the same, even though the ambient pressure has roughly doubled from its value at the surface.

 (d) To a first approximation, we expect the pressure difference to just be the hydrostatic pressure difference due the difference in elevation between the hoof and the head. Taking this to be 17 feet (5.2 m) and taking the density of giraffe blood to be 1000 kg/m^3, the pressure difference should be roughly $P = \rho g h$ = 1000 kg/m^3 * 9.8 m/s^2 * 5.2 m = 51,000 Pa = 0.5 bar. The pressure in the giraffe's hooves is about half an atmosphere higher than that in its head (when it is standing).

(e) It will fairly quickly increase by half an atmosphere, at least relative to the pressure in the giraffe's hooves. The actual increase is not quite so big, because the giraffe's heart moves to a slightly lower level when it stoops, and there accompanying changes in the giraffe's pulse rate, etc. that have some effect.

(f) The biggest problem would occur when the giraffe raises its head. Without special adaptations, the rapid change in pressure would drain all of the blood out of its head and it would faint (compare to the head rush you may sometimes get when changing the elevation of your head by just 3 feet or so). However, giraffes have a collapsible jugular vein and unusually strong ability to constrict the veins in their neck. This allows them to dramatically increase the flow resistance in their veins as they raise their heads, which prevents all of the blood from draining out. It seems that through differential contraction of different arterial paths, they can divert blood flow around the brain, or force a larger fraction through the brain. Overall, they have a much more sophisticated arrangement for controlling cranial blood pressure than most other animals.

1.17 (7th edition Prob. 1.14)

70 Watts times 10 hours per day is 700 Watt-hours per day or 0.7 kW-hour/day. Multiplying this by 365 days per year gives 255.5 kW-hours/year of electricity use at $0.10 per kW-hour, for a total annual electricity cost of $25.55. If the light is on 3650 hours per year, one will use an

average of 3.65 bulbs (each lasting 1000 hours, or 100 days at 10 hours per day) each year, at an average annual cost of $5*3.65 = $18.25/year. So the total annual cost to operate the light is about $18.25 + $25.55 = $44.

1.18 (7th edition Prob. 1.15)

(a) The force exerted on the gas has two parts, the weight of the piston and the pressure of the atmosphere on the top of the piston. The force due to the weight of the piston is mg/g_c where m is the piston mass and g is the acceleration of gravity. We have to use g_c to make the units work out right. The force due to the pressure of the atmosphere is PA where P is the pressure and A is the cross-sectional area of the piston. The cross-sectional area of the piston is $A = \pi d^2/4 = 0.7854*(1.25 \text{ ft})^2 = 1.227 \text{ ft}^2$. So, the total force is:

$$F = 250 \text{ lb}_m * 32.169 \text{ ft s}^{-2} / (32.1740 \text{ lb}_m \text{ ft lb}_f^{-1} \text{ s}^{-2}$$

$$+ 30.12 \text{ (in Hg)} * 0.49116 \text{ lb}_f \text{ in}^{-2} / \text{(in Hg)} * 1.227 \text{ ft}^2 * 144 \text{ in}^2 \text{ ft}^{-2}$$

$$F = 249.96 \text{ lb}_f + 2613.85 \text{ lb}_f = 2863.8 \text{ lb}_f$$

(b) The pressure of the gas is the force divided by the area

$$P = F / A = 2863.8 \text{ lb}_f / (1.227 \text{ ft}^2 * 144 \text{ in}^2 \text{ ft}^{-2}) = 16.21 \text{ lb}_f \text{ in}^{-2} = 16.21 \text{ psia}$$

(c) Because the weight of the piston, its cross-sectional area, and the pressure of the atmosphere on it all remain constant, the force on the gas and the gas pressure also remain constant. So, the work done by the gas can be computed as either $F \, \Delta h$ or $P \, \Delta V$, using the pressure and force that we have already computed. Since we are given $\Delta h = 1.7$ ft, it is easiest to calculate it as

$$W = 2863.8 \text{ lb}_f * 1.7 \text{ ft} = 4868 \text{ ft lb}_f$$

Most of this work is done against the pressure of the atmosphere, so only a small part of it changes the potential energy of the piston. The potential energy of the piston changes by

$mg/g_c\Delta h = 249.96 \ lb_f * 1.7 \ \text{ft} = 424.9 \ \text{ft lb}_f$

1.19 (7th edition Prob. 1.16)

(a) The force applied by the atmosphere (if the piston and weight were 'massless') would be the atmospheric pressure (101.57 kPa) times the cross-sectional area of the piston ($A = \pi d^2/4 = (0.47)^2\pi/4 \ \text{m}^2 = 0.173 \ \text{m}^2$. So, the force due to the atmosphere is 0.173 m^2 * 1.0157×10^5 N m^{-2} = 17600 N. The additional force applied by the piston and weight is $F = mg = 150 \ \text{kg} * 9.813 \ \text{m s}^{-2} = 1472$ N. Thus, the total force applied to the gas is about 19090 N.

(b) The pressure of the gas is the total force found in (a) divided by the cross sectional area of the piston. $P = 19090 \ \text{N} / 0.173 \ \text{m}^2 = 1.101×10^5 \ \text{N m}^{-2} = 1.101×10^5 \ \text{Pa} = 110.1$ kPa.

(c) Since the mass of the piston and the pressure of the atmosphere don't change in this process, the pressure of the gas inside the piston is constant as it expands, and the force applied by the piston and atmosphere to the gas is constant as the gas expands. We can compute the work either as the pressure times the change in volume or the force times the distance the piston moves. Since we've already computed the force in part (a), the simplest method is to multiply this force by the distance that the piston travels. $W = Fl = 19090 \ \text{N} * 0.83 \ \text{m} = 15850 \ \text{N m} = 15850$ J. The change in potential energy of the piston and weight is $\Delta E_P = mg\Delta z = 150 \ \text{kg} * 9.813 \ \text{m s}^{-2} * 0.83 \ \text{m} = 1222 \ \text{kg m}^2 \ \text{s}^{-2} = 1222$ J. Thus, most of the work done by the gas goes toward 'pushing back' the atmosphere, not raising the piston and weight. This is a result of the fact that most of the pressure being applied to the gas is due to the atmosphere, and not to the weight and piston.

1.20 (7th edition Prob. 1.17)

Kinetic energy is given by $E_K = \frac{1}{2} mv^2$. Its fundamental units are therefore the fundamental unit of mass (kg) times those of velocity (length/time, m/s):

$E_K \ [=] \ \text{kg} \cdot (\text{m/s})^2 \ [=] \ \text{kg} \cdot \text{m}^2/\text{s}^2 \ [=] \ \text{N} \cdot \text{m} \ [=] \ \text{J}$

Gravitational potential energy ($E_P = mgz$) has units of mass·length·acceleration. Its fundamental units are

therefore:

$$E_P \; [=] \; \text{kg·m·(m/s}^2) \; [=] \; \text{kg·m}^2/\text{s}^2 \; [=] \; \text{N·m} \; [=] \; \text{J}$$

1.21 (7th edition Prob. 1.18)

The kinetic energy is KE = ½ mv^2 = ½ * 1250 kg * (40 m/s)^2 = 1000000 kg m^2 s^{-2} = 1000000 J. To bring this car to a stop, a minimum of 1000000 J of work must be done on the car.

1.22 (7th edition Prob. 1.19)

We can either work forward, computing the energy output per mass of water, or work backward, starting from the energy requirement of the lightbulb. If we choose to work forward, then we compute that the potential energy change, per kilogram of water that falls is $\Delta E_P/m = g\Delta z = 9.8$ m s^{-2} *50 m = 490 m^2 s^{-2} = 490 J kg^{-1}. We are told that 91% of this energy is converted to electrical energy, and 92% of that electrical energy is transmitted to the lightbulb (8% is lost). So, the energy transmitted to the lightbulb is 490 J/kg * 0.91 * 0.92 = 410 J/kg. The 200-Watt light bulb requires 200 J per second. Dividing this by the energy production per kilogram gives

Required water flow rate = 200 J s^{-1} / 410 J/kg = 0.488 kg s^{-1} = 29.3 kg per minute = 64.5 lb$_m$ per minute.

To get more familiar volumetric units for water, we can divide by the density of water (about 8.3 lb$_m$ per gallon) to get 7.7 gallons per minute.

1.23 (New)

The kinetic energy per unit mass of the wind approaching the turbine is ½u^2 = ½ (12)^2 = 72 m^2 s^{-2} = 72 J/kg. The volumetric flow rate of air approaching the turbine is given by the velocity times

the cross-sectional area: $q = uA = 12$ m/s $* \pi (77$ m$)^2 /4 = 55880$ m^3/s. If the density of the air is 1.25 kg m^{-3}, then the mass flow rate is:

$$\dot{m} = \rho q = 1.25 \text{ kg m}^{-3} * 55880 \text{ m}^3 \text{ s}^{-1} = 69850 \text{ kg s}^{-1}$$

If this has a kinetic energy of 72 J kg^{-1}, then the total power available is 72*69850 = 5.03×10^6 J/s = 5.03 MW. The efficiency of the turbine (electrical energy output divided by kinetic energy of the approaching air) is 1.5/5.03 = 0.30.

1.24 (New)

(a) One way to figure this out is to find the annual average usage by taking the total annual usage (in kW-h) and dividing by the number of hours in a year to get an average usage in kW. For Buffalo, we have (6000 kW-h/yr)/(8766 h/yr) = 0.684 kW. In Pheonix, it is 11000/8766 = 1.255 kW. In Buffalo, we get 0.15*200 = 30 W m^2 of solar panel, while in Pheonix we get 0.15*270 = 40.5 W m^{-2}. The total area needed in Buffalo is therefore 684/30 = 22.8 m^2, while the total area needed in Pheonix is 1255/40.5 = 31.0 m^2.

(b) In Buffalo, it is 6000 kW h * 0.15 $/(kW h) = $900
In Pheonix, it is 11000 kW h * 0.09 $/(kW h) = $990

(c) Neglecting the time value of money, we need the solar panels in each case to cost less than 20 years worth of electricity. Thus, in Buffalo, we require the total cost to be less than 20*$900 = $18000 and in Pheonix we require it be less than 20*$990 = $19800. Dividing these by the total area needed in each location gives:

Buffalo: $18000/22.8 m^2 = $790 m^{-2}

Pheonix: $19800/31.0 m^2 = $639 m^{-2}

This must include the total installed cost in each case, not just the panels themselves. Support structures, power inverters, etc. will be required. Sizing the panels based on average insolation assumes that electricity can be stored and retrieved for indefinite periods of time at no additional cost. This might be the case if the local electricity supplier allows the homeowner to feed power

into the grid and be fully credited against any power drawn from the grid at other times (net metering).

1.25 (7th edition Prob. 1.20)

The exact conversion factors are:

1 atm = 1.01325 bar

1 Btu = 1.055 kJ

1 hp = 0.7457 kW

1 inch = 2.54 cm

1 lb_m = 0.45359 kg

1 mile = 1.6093 km

1 quart = 0.9464 liter

1 yard = 0.9144 m

Some other useful ones to remember are:

1 ft = 30.48 cm ≈ 30 cm = 0.3 m,

1 bar = 14.50 psi ≈ 15 psi, and

1 mile/hour = 0.447 m/s ≈ 0.5 m/s.

1.26 (7th edition Prob. 1.21)

There is a little ambiguity in the number of seconds per year, because of leap year, occasional addition of "leap-seconds", etc.

A reasonable number is 365.25 days*24 hours/day*60 min/hour*60 s/min = 31557600 s. Using this value, the above decimal calendar units are

1 Sc = 31.6 s = 0.53 minutes

1 Mn = 316 s = 5.26 minutes

1 Hr = 3156 s = 52.6 minutes = 0.877 hour

1 Dy = 526 minutes = 8.77 hour = 0.365 days

1 Wk = 87.7 hour = 3.65 days

1 Mo = 36.5 days

An inherent problem with this, or any other, decimal calendar is that a calendar needs to accommodate multiple "natural" time scales – at least the period of rotation of the earth, so that day/night cycles are reflected in the time units and the period of the earth's orbit around the sun, so that seasonal changes are reflected in the time units. Such "natural" scales that govern our everyday experience are less prevalent for other dimensions, such as mass, length, etc.

1.27 (7th edition Prob. 1.22).

(a) The cost of coal per GJ is $25/ton / (907.2 kg/ton * 0.029 GJ/kg) = $0.95 /GJ.

The cost of gasoline is $2/gal * 264.2 gal/m^3 / 37 GJ/m^3 = $14.30 /GJ.

The cost of electricity is $0.1 /(kW hr) / 0.0036 GJ/(kW hr) = $27.80 /GJ.

Coal is least expensive, electricity is most expensive.

(b) One can explain this in a variety of ways, but the underlying factor is that energy that is more convenient and/or more efficiently usable costs more. Electricity can be converted to mechanical energy (work) with nearly 100% efficiency (at least >90% in a well designed electric motor). In contrast, a gasoline-burning internal combustion engine might have a typical efficiency of 30 to 35%. Thus, with the costs in part (a), and all other things being equal, an electric car might be more cost effective to operate than a gasoline-powered one, because the higher efficiency of energy use would more than make up for the higher cost per unit of energy input. This is what makes electric or plug-in hybrid vehicles (potentially) economical.

On the other hand, coal is inexpensive because it is very inconvenient to use. It cannot readily be used on a small scale, and is mostly consumed in large-scale industrial electricity generation and heating processes. A coal-fired power plant can convert about 1/3 of the energy of the coal into electricity. If that were the only cost of generating electricity, then with the costs in part (a), generating electricity from coal would be a very profitable business. However, the *consumer*

electricity cost of \$0.10/kW*hr reflects (along with profits) the costs of building the coal-fired power plant (although most US plants are old, and fully amortized) the cost of crushing and cleaning up the coal for combustion, the cost of pollution mitigation, and the cost of transmitting the power from the power plant . In mid-2008, the actual average coal cost for power generation was about \$2 /GJ, which leads to an electricity generation cost of something like \$0.03 per kW-hr, without including transmission costs, costs of pollution/cleanup, or capital costs of building the plant.

1.28 (7th edition Prob. 1.23)

(a) The cost per unit size is $C/S = \alpha S^{\beta}/S = \alpha S^{\beta-1}$. If β is less than 1, then the exponent $(\beta - 1)$ is negative, and the cost per size decreases with increasing size (S).

(b) To a good first approximation, the cost of the tank should be proportional to the amount (mass) of material required to construct the tank.

If the wall thickness, t, is independent of the tank volume, and is small compared to the tank diameter, then the mass of material required to make the tank will be $m = \rho * t * \pi d^2$, where d is the diameter of the spherical tank and ρ is the density of the material from which it is constructed.

The tank volume is $V^t_t = \pi d^3/6$.

Or, writing the diameter in terms of the volume, $d = (6V^t_t/\pi)^{1/3}$.

Substituting this into the expression for the mass of material gives

$$m = \rho * t * \pi * (6V^t_t/\pi)^{2/3} = \rho * t * (36/\pi)^{1/3} \, V^{t\,2/3}_t$$

This shows that the exponent should be approximately $\beta = 2/3$. It also shows that the cost should depend on the thickness of the tank walls, which will, in turn, depend on the maximum pressure that the tank must be able to withstand. Other factors that will influence the quantity α should include the material of construction and the fabrication techniques used. The material of construction will depend on the need for corrosion resistance and other such factors.

1.29 (7th edition Prob. 1.24)

This can be done using many different software packages. The Microsoft Excel spreadsheet shown below illustrates one way of doing such nonlinear regression:

A	B	C
12.74	2017.87	-80.87

T (°C)	T(K)	P^{sat} (kPa)	ln (P^{sat})	ln (P^{sat})$_{fit}$	Error^2	P^{sat}_{fit} (kPa)	Relative Error
-18.5	254.65	3.18	1.157	1.125	1.02E-03	3.08	-3.2%
-9.5	263.65	5.48	1.701	1.697	2.03E-05	5.46	-0.4%
0.2	273.35	9.45	2.246	2.253	4.77E-05	9.52	0.7%
11.8	284.95	16.9	2.827	2.849	4.61E-04	17.27	2.2%
23.1	296.25	28.2	3.339	3.368	7.96E-04	29.01	2.9%
32.7	305.85	41.9	3.735	3.767	1.02E-03	43.26	3.3%
44.4	317.55	66.6	4.199	4.211	1.43E-04	67.40	1.2%
52.1	325.25	89.5	4.494	4.479	2.24E-04	88.17	-1.5%
63.3	336.45	129	4.860	4.841	3.50E-04	126.61	-1.9%
75.5	348.65	187	5.231	5.201	9.19E-04	181.42	-3.0%

Error Sum   5.01E-03
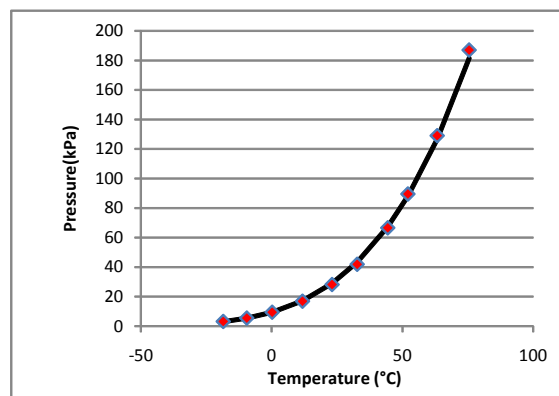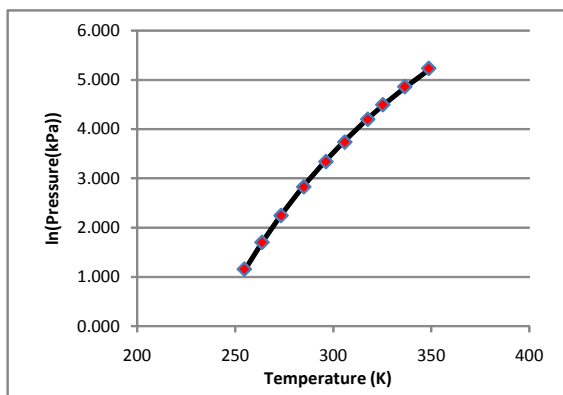
The first and third columns have the data given in the problem statement. The second column has the temperature in K, and the 4th column has the natural logarithm of the vapor pressure data. The fifth column has the result of

$$\ln P^{sat} / \mathrm{kPa} = A - \frac{B}{T/\mathrm{K}+C}$$

With parameters A, B, and C as shown at the top of the spreadsheet. The sixth column has the square of the difference between the actual and model values of $\ln(P^{sat})$ (column 5 – column 4)^2. The cell labeled "Error Sum" has the sum of those differences. The Solver function in MS Excel was used to minimize that cell (the sum of the squares of the differences between the data and fit values) by simultaneously varying the values of A, B, and C.

The 6th column has the vapor pressure predicted by the fit, and the 7th column has the relative error in the vapor pressure at each point, which is (fit value – actual value)/actual value. The two plots show the data and fit in two formats, in each case with the data as discrete points and the fit as a smooth curve. The one on the left is what is actually used in the fitting. That is, the sum of

the squares of the distances between the points and the curve was minimized to obtain the parameters. The one on the right shows the data and fit in the original format in which the data was given.

The boiling point is the temperature at which $P^{sat} = 101.325$ kPa. If one already has the spreadsheet shown above, then the simplest way to determine this is to vary one of the temperatures until the fitted pressure is equal to 101.325 kPa. This gives a value of 56.3 °C.